

INTERNAL COMBUSTION ENGINE

Module – IV

Working cycle:

(a) *Otto cycle*- thermodynamic cycle for SI/petrol engine

-Reversible adiabatic compression and expansion process

-Constant volume heat addition (combustion) and heat rejection process (exhaust)

Figure 7 depicts the Otto cycle

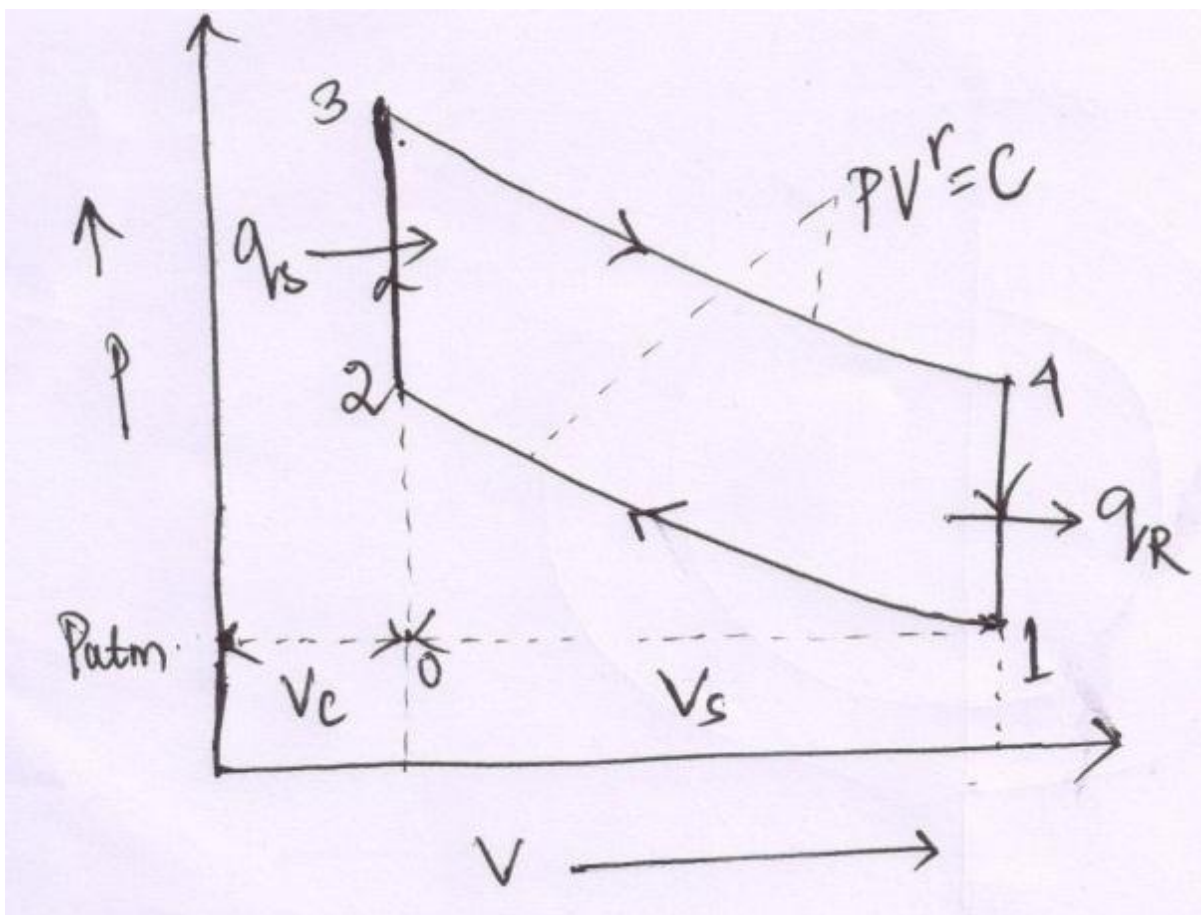


Fig. 7. Otto cycle

Heat supplied, $q_s = C_v(T_3 - T_2)$

Heat rejection, $q_R = C_v(T_4 - T_1)$

Compression ratio, $r_k = \frac{V_1}{V_2}$

$$\text{Thermal efficiency, } \eta_{th} = \frac{q_s - q_R}{q_s} = \frac{Cv(T_3 - T_2) - Cv(T_4 - T_1)}{Cv(T_3 - T_2)} = 1 - \frac{T_4 - T_1}{T_3 - T_2}$$

In process 1-2, adiabatic compression process,

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\gamma-1}$$

$$\Rightarrow T_2 = T_1 \cdot (r_k)^{\gamma-1}$$

In adiabatic expansion process, i.e. 3-4,

$$\frac{T_4}{T_3} = \left(\frac{V_3}{V_4}\right)^{\gamma-1} = \left(\frac{V_2}{V_1}\right)^{\gamma-1}$$

$$\Rightarrow T_3 = T_4 \cdot (r_k)^{\gamma-1}$$

$$\eta_{th} = 1 - \frac{T_4 - T_1}{T_4 \cdot (r_k)^{\gamma-1} - T_1 \cdot (r_k)^{\gamma-1}}$$

$$= 1 - \frac{1}{(r_k)^{\gamma-1}}$$

Work done (W)

Pressure ratio, $r_p = \frac{P_3}{P_2} = \frac{P_4}{P_1}$

$$\frac{P_2}{P_1} = \frac{P_3}{P_4} = \left(\frac{V_1}{V_2}\right)^\gamma = (r_k)^\gamma$$

$$\begin{aligned} W &= \frac{P_3 V_3 - P_4 V_4}{\gamma - 1} - \frac{P_2 V_2 - P_1 V_1}{\gamma - 1} \\ &= \frac{1}{\gamma - 1} \left[P_4 V_4 \left(\frac{P_3 V_3}{P_4 V_4} - 1 \right) - P_1 V_1 \left(\frac{P_2 V_2}{P_1 V_1} - 1 \right) \right] \\ &= \frac{1}{\gamma - 1} [P_4 V_1 (r_k)^{\gamma-1} - 1 - P_1 V_1 (r_k)^{\gamma-1} - 1] \\ &= \frac{P_1 V_1}{\gamma - 1} [r_p (r_k)^{\gamma-1} - 1 - (r_k)^{\gamma-1} - 1] \\ &= \frac{P_1 V_1}{\gamma - 1} [(r_k)^{\gamma-1} - 1] (r_p - 1) \end{aligned}$$

$$\text{Mean effective pressure, } P_m = \frac{\text{work done}}{\text{swept volume}} = \frac{\text{work done}}{V_1 - V_2}$$

$$P_m = \frac{\frac{P_1 V_1}{\gamma - 1} [(r_k^{\gamma-1} - 1)(r_p - 1)]}{V_1 - V_2} = \frac{P_1 r_k [(r_k^{\gamma-1} - 1)(r_p - 1)]}{(\gamma - 1)(r_k - 1)}$$

(b) Diesel cycle- thermodynamic cycle for low speed CI/diesel engine

- Reversible adiabatic compression and expansion process
- Constant pressure heat addition (combustion) and heat rejection process (exhaust)

Figure 8 depicts the diesel cycle.

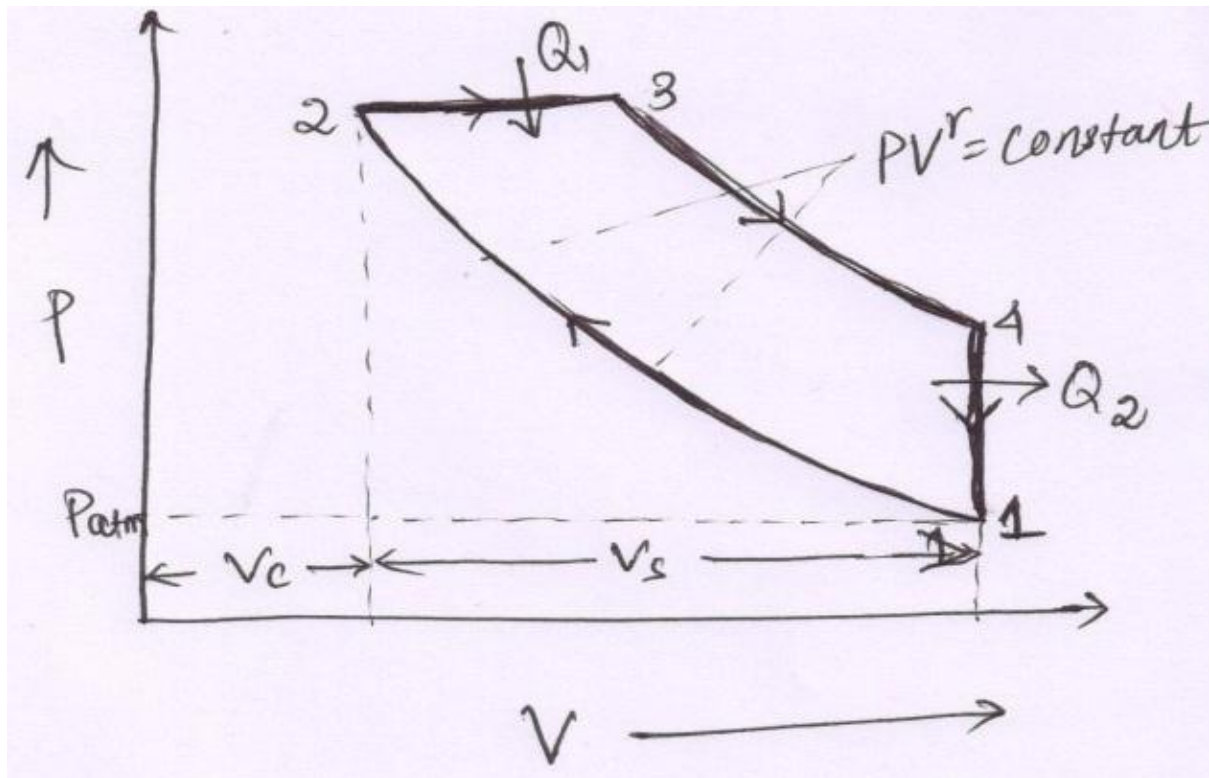


Fig. 8. Diesel cycle

Heat supplied, $Q_1 = C_p(T_3 - T_2)$

Heat rejection, $Q_2 = C_v(T_4 - T_1)$

Compression ratio, $r_k = \frac{V_1}{V_2}$

Cut off ratio, $r_c = \frac{V_3}{V_2}$

Thermal efficiency, $\eta_{th} = \frac{Q_1 - Q_2}{Q_1} = \frac{C_p(T_3 - T_2) - C_v(T_4 - T_1)}{C_p(T_3 - T_2)} = 1 - \frac{1}{\gamma} \frac{(T_4 - T_1)}{(T_3 - T_2)}$

In adiabatic compression process i.e. 1-2,

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\gamma-1}$$

$$\Rightarrow T_2 = T_1 \cdot (r_k)^{\gamma-1}$$

In process 2-3, pressure constant, then

$$\frac{T_3}{T_2} = \frac{V_3}{V_2} = r_c$$

$$\Rightarrow T_3 = T_2 \cdot r_c = T_1 \cdot (r_k)^{\gamma-1} \cdot r_c$$

In adiabatic expansion process i.e. 3-4,

$$\frac{T_4}{T_3} = \left(\frac{V_3}{V_4}\right)^{\gamma-1} = \left(\frac{V_3}{V_2} \cdot \frac{V_2}{V_4}\right)^{\gamma-1} = (r_c)^{\gamma-1} \cdot \frac{1}{(r_k)^{\gamma-1}}$$

$$\Rightarrow T_4 = T_3 \cdot (r_c)^{\gamma-1} \cdot \frac{1}{(r_k)^{\gamma-1}} = T_1 \cdot (r_k)^{\gamma-1} \cdot r_c \cdot (r_c)^{\gamma-1} \cdot \frac{1}{(r_k)^{\gamma-1}} = T_1 \cdot r_c$$

$$\eta_{th} = 1 - \frac{1}{\gamma} \frac{(T_4 - T_1)}{(T_3 - T_2)} = 1 - \frac{1}{\gamma \cdot (r_k)^{\gamma-1}} \left[\frac{(r_c)^{\gamma} - 1}{r_c - 1} \right]$$

Work done (W)

$$W = P_2(V_3 - V_2) + \frac{P_3 V_3 - P_4 V_4}{\gamma - 1} - \frac{P_2 V_2 - P_1 V_1}{\gamma - 1}$$

$$= P_2(r_c V_2 - V_2) + \frac{P_2 r_c V_2 - P_4 r_k V_2}{\gamma - 1} - \frac{P_2 V_2 - P_1 r_k V_2}{\gamma - 1} \quad \text{since } V_4 = V_1$$

$$= P_2 V_2 \left[\frac{(r_c - 1)(\gamma - 1) + (r_c - r_c^{\gamma} r_k^{-\gamma} r_k) - (1 - r_k^{1-\gamma})}{\gamma - 1} \right]$$

$$= P_1 V_1 \cdot r_k^{\gamma-1} \left[\frac{\gamma(r_c - 1) - r_k^{1-\gamma}(r_c^{\gamma} - 1)}{\gamma - 1} \right]$$

Mean effective pressure,

$$P_m = \frac{P_1 V_1 r_k^{\gamma-1} \left[\frac{\gamma(r_c-1) - r_k^{1-\gamma} (r_c^{\gamma}-1)}{\gamma-1} \right]}{V_1 - V_2} = \frac{P_1 r_k^{\gamma} [\gamma(r_c-1) - r_k^{1-\gamma} (r_c^{\gamma}-1)]}{(\gamma-1)(r_k-1)}$$

(c) Dual cycle or limited pressure cycle-thermodynamic cycle for high speed diesel and hot spot ignition engine

- Reversible adiabatic compression and expansion process
- Constant pressure and constant volume heat addition (combustion) and heat rejection process

Total heat supplied, $Q_1 = C_v(T_3 - T_2) + C_p(T_4 - T_3)$

Heat rejection, $Q_2 = C_v(T_5 - T_1)$

Compression ratio, $r_k = \frac{V_1}{V_2}$

Cut off ratio, $r_c = \frac{V_4}{V_3}$

Pressure ratio, $r_p = \frac{P_3}{P_2}$

Figure 9 shows the P-V diagram of Dual cycle.

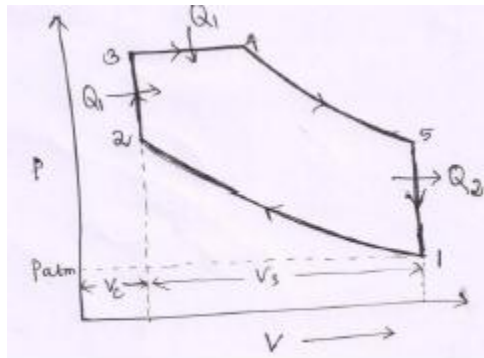


Fig. 9. Dual cycle

$$\text{Thermal efficiency, } \eta_{th} = \frac{Q_1 - Q_2}{Q_1} = \frac{C_v(T_3 - T_2) + C_p(T_4 - T_3) - C_v(T_5 - T_1)}{C_v(T_3 - T_2) + C_p(T_4 - T_3)} = 1 - \frac{(T_5 - T_1)}{(T_3 - T_2) + \gamma(T_4 - T_3)}$$

In adiabatic compression process i.e. 1-2,

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2} \right)^{\gamma-1} = (r_k)^{\gamma-1}$$

In constant volume combustion process i.e. 2-3,

$$\frac{P_3}{P_2} = \frac{T_3}{T_2} = r_p$$

$$\Rightarrow T_2 = \frac{T_3}{r_p}$$

In constant pressure combustion process i.e. 3-4,

$$\frac{V_3}{V_4} = \frac{T_3}{T_4}$$

$$\Rightarrow T_4 = T_3 \cdot r_c$$

In adiabatic expansion process i.e. 4-5,

$$\frac{T_4}{T_5} = \left(\frac{V_5}{V_4}\right)^{\gamma-1} = \left(\frac{V_1}{V_4}\right)^{\gamma-1} = \left(\frac{r_k}{r_c}\right)^{\gamma-1}$$

$$\Rightarrow T_5 = r_c \cdot T_3 \cdot \left(\frac{r_c}{r_k}\right)^{\gamma-1}$$

$$\eta_{th} = 1 - \frac{(T_5 - T_1)}{(T_3 - T_2) + \gamma(T_4 - T_3)} = 1 - \frac{1}{(r_k)^{\gamma-1}} \left[\frac{r_p \cdot (r_c)^{\gamma-1}}{(r_p - 1) + \gamma r_p (r_c - 1)} \right]$$

Work done (W)

$$\begin{aligned} W &= P_2(V_4 - V_2) + \frac{P_4V_4 - P_5V_5}{\gamma - 1} - \frac{P_2V_2 - P_1V_1}{\gamma - 1} \\ &= P_2V_2(r_c - 1) + \frac{(P_4r_cV_2 - P_5r_kV_2) - (P_2V_2 - P_1r_kV_2)}{\gamma - 1} \\ &= \frac{P_1V_1 \cdot r_k^{\gamma-1} [\gamma r_p (r_c - 1) + (r_p - 1) - r_k^{\gamma-1} (r_p r_c^{\gamma} - 1)]}{\gamma - 1} \end{aligned}$$

Mean effective pressure,

$$P_m = \frac{P_1 V_1 \cdot r_k^{\gamma-1} [\gamma r_p (r_c - 1) + (r_p - 1) - r_k^{\gamma-1} (r_p r_c^\gamma - 1)]}{V_1 - V_2}$$
$$= \frac{P_1 r_k^\gamma [r_p (r_c - 1) + (r_p - 1) - r_k^{1-\gamma} (r_p r_c^\gamma - 1)]}{(\gamma - 1)(r_k - 1)}$$

Comparison of Otto, Diesel and Dual cycle:

(a) For same compression ratio and same heat input

$$(\eta_{th})_{Otto} > (\eta_{th})_{Dual} > (\eta_{th})_{Diesel}$$

(b) For constant maximum pressure and same heat input

$$(\eta_{th})_{Diesel} > (\eta_{th})_{Dual} > (\eta_{th})_{Otto}$$

(c) For same maximum pressure and temperature

$$(\eta_{th})_{Diesel} > (\eta_{th})_{Dual} > (\eta_{th})_{Otto}$$

(d) For same maximum pressure and output

$$(\eta_{th})_{Diesel} > (\eta_{th})_{Otto}$$